| AD | S/B | Subject | EO | Statement of compliance |
|--|---|--|-------------|---|
| 1973-01-03, 1983-01-06 | Air Cruisers S/B 112- 72-1 Switlik Parachute Company, Inc., Service Bulletin 25- 00-16 | Inspect - Crew and Supernumerary Life Vest P/N | 25-80190-02 | Accomplishing this EO verifies compliance with ADs listed in the "Referenced Documents" section of this EO. |
| 1974-08-09 | | Inspect - Lavatory Waste Compartment | 25-80100-03 | This EO accomplishes the initial and repetitive inspections of the Forward Lavatory Receptacles, Placards, and Ashtrays I/A/W paragraphs (g), (i) and (k) of AD 74-08-09, R3, amendment 39-16951 dated 3/28/2012. |
| 1979-18-05 | | Inspect - LiSO2 Batteries | 25-80216-01 | Completion of this EO will verify the aircraft meets the requirement of AD 79-18-05, amendment 39-3708, dated 02/28/1980. |
| 1980-01-05 | Pacific Scientific S/B 1101550-25-11 | Inspect - PSC Seat Restraint Buckles | 25-80193-02 | Completion of this EO will verify if aircraft meets the intent of AD 80-01-05, R1, amendment. 39-3851, dated 7/10/1980 and Pacific Scientific SB 1101550-25-11, dated 8/2/1979. |
| 1984-18-07 | | Inspect - Fire Extinguisher Discharge Cartridge for Presence of Aluminum Foil | 26-80208-01 | This EO verifies compliance with AD 84-18-07, amendment 39-4915, dated 09/21/1984 paragraph (a). |
| 1987-06-09 | | Inspect - Mechanical Product Circuit Breakers | 24-80194-02 | Accomplishment of this EO along with previous operators compliance documentation, will ensure the requirements mandated by AD 87-06-09, amendment 39-5585, effective date 04/ 27/1987 paragraph (a) are met. |
| 1987-08-09 | | Inspect/Install - Tire Servicing Placard | 12-80264-01 | This EO complies with paragraph (a)(1) of AD 87-08-09, amendment 39-5613 dated 06/01/1987. |
| 1988-19-05 | 767-54A0028 | Inspect - Nacelle Firewall Seal Installation | 54-80250-01 | This EO verifies compliance with paragraph (A) of AD 88-19-05, amendment 39-6002 dated 09/26/1988. |
| 1990-09-10 | 767-24A0060 | Inspect/Replace -Three-Phase Circuit Breakers | 24-80227-01 | This task complies with paragraph (a)(1) of AD 1990-09-10, amendment 39-6565, effective 05/30/1990. |
| 1990-12-02 (a), (b) | Hamilton Standard S/B 36-2078 Hamilton Standard S/B 36-2116 | Inspect - Intermediate Pressure Check Valve | 36-80232-01 | Accomplishment of this EO will verify compliance with paragraphs (a),(b) of AD 1990-12-02 amendment 39- 6620, effective 06/18/1990. AD 1990-12-02 requires repetitive inspections of a Hamilton Standard Intermediate Pressure Check Valve IAW S/B 36-2078. FAA letter 92-130S-1205, dated 11/24/1992 approves a one time inspection and re- identification of the subject check valves IAW H.S. S/B 36-2116 as an AMOC and terminating action to the requirements of AD 1990-12-02. |
| 1990-25-08 | 767-25-0193 | Inspect - Firestop Installation FWD and AFT Lower Cargo Compartments | 25-80186-01 | Compliance with this EO verifies compliance with SB 767-25-0193 Rev 01, Dated 12/21/1992 as mandated by AD 90 25-08, Amendment 39-6816, dated 01/02/1991. The FAA has approved S/B 767-25-0193, Rev. 01 as an alternative to the requirements specified in Airworthiness Directive 90-25-08. |
| 1991-10-01, 1992-11-09, 1994-01-04, 1994-01-05, 1994-01-06, 1994-02-02, 1999-10-14, 2002-06-06, 2005-18-09, 2012-02-08, 2014-05-27 | | Inspect - Avionics Equipment | 34-80201-04 | This EO verifies compliance with ADs as listed in "Reference Documents" section by verifying the affected equipment is NOT installed on the aircraft. |
| 1991-15-15 | PELS PBE Service | Inspect - Crewmember Protective Breathing | 35-80195-01 | Compliance with this EO verifies there are no crewmember PBE units on the aircraft that are |
| 1993-09-01 | Bulletin No. 002 767-26A0089 | Equipment (PBE) Inspect/Replace - Fire Extinguisher Discharge Cartridge | 26-80200-01 | affected by AD 91-15-15 AMND 39-7079, Dated 08/15/1991. This task complies with the intent of Boeing S/B 767-26A0089, IR, dated 10/22/1992 as amended by Notice of Status Change 767-26A0089 NSC 1, 2 and 3 as mandated by AD 93-09-01, amendment 39-8568 effective 06/09/1993 paragraphs (a), (b), (c) and (g). |

| AD | S/B | Subject | EO | Statement of compliance |
|--|--------------------------------|---|---------------------|---|
| 1993-13-01 | 767-24A0085 | Inspect - Static Inverter | 24-80203-01 | This EO verifies that Boeing SB 767-24A0085 Rev. 0, Dated September 17, 1992 mandated by AD 93-13-01 Amendment 39-8614, Dated August 9, 1993 has been accomplished. |
| 1993-14-19 | 767-57A0101 | Inspect/Replace - Leading Edge Devices, Slat Trailing Edge Wedge | 57-80158-05 | This EO will accomplish repetitive inspections and if necessary, repairs of the leading edge slat trailing edge wedges. Inspections and repairs are accomplished IAW S/B 767-57A0101 original issue, dated 1/27/2005 as mandated by AD 93-14-19, amendment 39-8644, dated 9/2/1993, as well as data included in 767-57A0101-AMOC-01. AD 93-14-19, Amdt. 39-8644 mandates repetitive inspections of the slat trailing edge wedges. Installation of a current production or PAA core trailing edge wedge IAW S/B 767-57A0101 terminates repetitive inspections mandated by AD 93-14-19 for an individual slat. ** This EO is currently applicable to SB Group 1 and SB Group 2 aircraft. The only difference is the compliance times. Work will be accomplished Before Further Flight which overrides the compliance times. |
| 1993-24-16, 1997-11-10, 1997-18-03, 2009-21-10, 1981-08-04 | | Inspect - O2 System Components | 35-80202-03 | This EO verifies compliance with the following ADs: AD 81-08-04, amendment 39-4092, effective date 04/23/1981 AD 93-24-16, amendment 39-8765, effective date 02/04/1994 paragraph (a) AD 97-11-10, amendment 39-10039, effective 06/23/1997 paragraph (b) AD 97-18-03, amendment 39-10113, effective 09/22/1997 paragraph (b) AD 2009-21-10 amendment 39-16049 effective 02/04/2010 paragraph (g). |
| 1994-03-07 | 767-32-0132 | Inspect - Brake Disconnects, B767 | 32-80223-01 | This EO verifies that the aircraft brake disconnect assemblies have been modified by the approved AMOC Boeing SB 767-32-0132, IR. Dated 06/29/1995 to comply with par (b)(1) of AD 94-03-07 Amendment 39-8814, Ef fective 02/18/1994. |
| 1994-16-03 | 767-78-0060 | Test - Thrust Reverser Synch-Lock, CF6-80A | 78-80244-01 | Accomplishment of this EO complies with the repetitive tests required by paragraph (e) of AD 1994-16-03. |
| 1994-21-06 | 1108435-25-01 1108460-25-01 | Inspect - Pacific Scientific Lap Belt Assembly | 25-80204-01 | This EO verifies compliance as mandated by AD 94-21-06, amendment 39-9048, effective Nov. 25, 1994 paragraphs (a), (b), and (e). |
| 1995-01-01 | | Inspect - Terra Corporation TRT 250 Series Transponders | 34-80238-01 | This EO complies with AD 1995-01-01 paragraphs (a), (b) and (d). |
| 1995-10-11 (c) | Martin Marietta S/B 78-1002 | Inspect/Verify - Thrust Reverser, CDU, P/N & S/N | 78-80228-01 | This EO verifies compliance with Martin Marietta S/B 78-1002 as mandated by AD 1995-10-11, amendment 39- 9228, effective date 05/31/1995, paragraph (c). |
| 1996-07-09 | | EICAS Computer and Software Upgrade | ADV-31-80161- 01 | Compliance with this ADV will verify the accomplishment of the intent of AD 96-07-09, amendment 39-9558, dated 05/03/1996, paragraph (d)(1), (EICAS computer upgrade) and the intent of AD 2011-25-11, amendment 39-16887, dated 01/17/2012. paragraph (i)(2), (OPS Version 7 upgrade) and (j)(8)&(12), (EICAS computer and OPC upgrade). S8 767-31-295, IR, dated 06/16/2014 further upgrades the Operational Program Software (OPS) to Version 8. S8 767-31-295 has been approved as an AMOC as an alternative method of compliance to the requirements of paragraph (d)(1) of AD 96-07-09 and as an alternative method of compliance to the requirements of paragraph (i)(2) of AD 2011-25-11 |
| 1996-14-51 | 767-24A0113 | Inspect - Aileron Control Cable & Power Feeder Cable - Lower Lobe | 24-80179-02 | Compliance with this task accomplishes the intent of SB 767-24A0113, Rev 02, dated 10/17/1996 as mandated by AD 96-14-51, amendment 39-9695, effective 07/22/1996 paragraphs (a) and (b). S/B 767- 24A0113, Rev 2, dated 10/17/1996 has been approved as an AMOC to the requirements of AD 96-14-51. |
| 1996-19-10 | 767-24A0112 | Inspect/Replace-Main Battery Shunt & Main Battery Ground Stud Connection | 24-80180-02 | Accomplishment of this task complies with Boeing S/B 767-24A0112, Rev 2, dated 10/24/1996, as mandated by AD 96-19-10, amendment 39-9757, effective 10/02/1996 paragraph (a). S/B 767-24A0112, Rev 1, dated 08/01/1996 has been approved as an AMOC to the requirements of AD 96-19-10. |
| 1996-20-05 | 767-29A0080 | Inspect - Ram Air Turbine (RAT) Deployment | 29-80243-01 | Accomplishment of this EO complies with the intent of S/B 767-29A0080, Rev 02 dated 03/23/2000 as mandated by paragraph (a) of AD 1996-20-05, amendment 39-9771 dated 11/04/1996 S/B 767-29A0080, Rev 02 has been approved as an AMOC to the requirements of paragraph (a) of AD 96- 20-05. |

| AD | S/B | Subject | EO | Statement of compliance |
|------------|---|--|-------------|--|
| 1997-15-09 | 767-24A0111 | Verify/Install - APU Dual Ground | 24-80181-01 | Accomplishing this EO verifies compliance with SB 767-24A0111, Rev 01, Dated 10/30/1997 as mandated by AD 97-15-09 Amendment 39-10083, Dated 08/01/1997. |
| | | | | Repeat inspections required by AD 97-15-09 were terminated by AD 97-15-09 AMOC 97-130S-791 and AD 97-15-09 AMOC 97-130S-969. Accomplishment of this EO complies with Boeing S/B 767-32A0157, Rev0, dated 10-10-96, as mandated by AD 97- |
| 1997-16-03 | 767-32A0157 | Inspect/Verify - MLG & Doors-Forward Trunnion Joint Retaining Bolt | 32-80182-02 | Id-03, AMND 39-10093, Dated 9-5-97, paragraph (a). Compliance with this task accomplishes the intent of SB 767-25-0244 Rev 02, Dated 09/02/2004 as mandated by |
| 1998-03-10 | 767-25-0244 | Inspect - Captains and F/O's Seat Track | 25-80177-01 | AD 98-03-10 amendment 39-10302, effective 02/19/1998 paragraphs (a) and (b). |
| 1998-07-26 | 767-24A0120 | Inspect - Electrical Wire Bundles | 24-80238-01 | This EO complies with the intent of Boeing Message Number M-7200-98-00140, dated January 11, 1998. as mandated by paragraph (a) of AD 1998-07-26 amendment 39-10448 dated 04/21/1998. |
| 1998-10-10 | | Verify - Boost Pump Wiring Sleeve | 28-80206-01 | This task complies with the intent of Boeing Message M-7200-98-01080 dated March 18, 1998 as mandated by AD 98-10-10, amendment 39-10522 effective 5/27/1998 paragraphs (a), (b)(1)(2)(i)(ii), (c),and (f). |
| 1999-24-02 | 767-32A0185 | Inspect - Nose Landing Gear - Lower Drag Strut | 32-80176-02 | Accomplishment of this task complies with Boeing S/B 767-32A0185, Rev00, dated 09/02/1999, as mandated by paragraphs (a), (b), (c), (d) and (e) of AD 1999-24-02, amendment 39-11426, dated 12/06/1999. |
| 2000-03-21 | 767-32A0163 | Verify-Wire Bundle Protect, MLG Control Lever | 32-80174-01 | Completion of this task accomplishes the intent of Service Bulletin 767-32A0163 Rev 01 Dated 10/01/1998, as mandated by AD 2000-03-21 Amendment 39-11581, Dated 03/28/2000. |
| 2000-09-04 | 767-78-0063 | Inspect - Thrust Reverser Electro- Mechanical/CDU Cone Brake | 78-80102-03 | Compliance with this EO accomplishes the 1,000 hour repetitive tests IAW Appendix 1 of AD 2000-09-04 as mandated by AD 2000-09-04, amendment 39-11712 dated 5/19/00, paragraphs (e), (f) & (g). |
| 2000-16-14 | 767-33A0075 | Inspect - Lower Body Anti-collision Light, Cable Assembly | 33-80221-02 | This task complies with the intent of Boeing S/B 767-33A0075, Rev 02, dated 09/24/2009 as mandated by AD 2000- 16-14, amendment 39-11871, effective 09/25/2000 paragraph (a). S/B 767-3A0075 Rev. 02 has been approved as an AMOC to the requirements of paragraph (a) of AD 2000-16-14, amendment 39-11871, effective 09/25/2000. Additionally AMOC letter 130S-00-0613, dated 10/12/2000 allows for additional Connector P/Ns to be installed on the Anit-Collision Light Cable |
| 2000-17-06 | 767-78-0084 | Check - Thrust Reverser Directional Pilot Valve | 78-80103-02 | This EO complies with the repetitive actions of S/B 767-78-0084, Rev. 4, dated 12/6/2007, as mandated by AD 2000-17-16, amendment 39-11880 dated 10/05/2000 paragraph (b). |
| 2001-11-08 | 767-31-0106 | Verify - Warning Electronics Unit (WEU) Power Supply | 31-80199-01 | S/B 767-78-0084, Rev.4 has been approved as an alternative method of compliance to the requirements of This EO verifies incorporation of SB 767-31-0106 Rev 02 as mandated by AD 2001-11-08 AMND 39-12250, effective 07/16/2001 paragraphs (b), (c)(3), (e)(1)(2), (f), and (g). |
| 2001-12-03 | 767-33A0085-03 | Inspect - Passenger Compartment Light - Wire Bundles | 33-80175-02 | Accomplishment of this task complies with Boeing S/B 767-33A0085-03, dated 10/31/2002, as mandated by AD 2001-12-03, amendment 39-12258, effective 07/18/2001 paragraph (a). |
| 2001-16-03 | 767-78A0090 | Inspect/Check - Engine, Thrust Reverser System | 78-80104-01 | This EO accomplishes the repetitive inspections and tests of the Thrust Reverser Control System and any corrective action required as specified in Boeing Service Bulletin 767-78A0090, Rev 04 dated 09/30/2004 as mandated by AD 2001-16-03, amendment 39-12371 dated 08/21/2001 paragraphs (a) and (b). S/B 767-78A0090, Rev 04 dated 09/30/2004 has been approved as an AMOC to the requirements of AD 2001-16-03 paragraphs (a) and (b). |
| 2001-16-07 | 767-71-0088 Rohr S/B TBC/80C2- NAC-71-028 | Inspect - P/N, Engine Core Cowl | 71-80254-01 | This EO verifies compliance with Boeing S/B 767-71-0088 as mandated by paragraph (a) of AD 2011-16-07 amendment 39-12375, effective September 24, 2001. |
| 2001-18-12 | 767-24A0139 767-24A0144 | Inspect - Electrical Power Wire Bundle | 24-80162-04 | Compliance with this EO and JC24-80162-03-001 will accomplish the intent of Service Bulletin SB 767- 24A0139, Rev. 1 dated Jan. 30, 2003 as mandated by AD 2001-18-12 paragraphs (a) and (b). S/B 767- 24A139, Rev. 1 has been approved as an AMOC to the requirements of AD 2001-18-12 paragraphs (a) and (b). A 6,000 FH or 18 months WOF repeat inspection is required. Compliance with this EO and JC24-80162-03-101 will accomplish the intent of Service Bulletin SB 767-24A01144, Rev. 1 dated 11/04/2004. Service Bulletin SB 767-24A01144, Rev. 1 dated 11/04/2004 has been approved as an AMOC to the requirements of AD 2001-18-12 para (a) and (b) and is terminating action to the repetitive inspections required by S/B 767-24A0139. |

| AD | S/B | Subject | EO | Statement of compliance |
|-----------------------------------|---|---|-------------|--|
| 2001-22-13 and 98-13-12 | 767-27A0159 Rev 01, dated 04/05/2001 | Inspect/Replace - Rudder Pedal Pushrod Fasteners | 27-80197-02 | This task incorporates Service Bulletin 767-27A0159 Rev 01, dated 04/05/2001 as mandated by paragraph (a) of AD 2001-22-13, amendment 39-12492, effective 12/06/2001. This task also complies with the requirements of AD 98-13-12R1 as specified in paragraph (e) of AD 2001-22-13. |
| 2002-22-12 | Titeflex S/B 73-2 | Inspect/Replace - Titeflex Hoses, CF6-80C2B6 | 72-80196-01 | Compliance with EO accomplishes the intent of Titeflex corporation SB 73-2, rev IR, dated 11/27/2000 as mandated by paragraphs (a) and (b) of AD 2002-22-12 amendment 39-12938, effective 12/13/2002 as applicable for hose installations on GE CF6-80C2B6 engines. |
| 2003-14-09 | 767-57A0066 | Inspect/Replace Left and Right Inboard Flap Splined Components | 57-80105-04 | This EO accomplishes the removal, inspection and replacement of the Splined components of the Left and Right Inboard Flap (Collar Fittings, T orque Tube, and Bushings) in accordance with Service Bulletin 767-57A0066, Rev. 4, dated Feb. 27, 2012. as mandated by Airworthiness Directive 2003-14-09, Amendment 39-13228, dated 08/22/2003; supersedes Airworthiness Directive 98-22-12, amendment 39-10859, dated 11/12/1998. |
| 2003-18-10 | 767-53-A260-00 | Inspect - Sect 41 - S-2R Lap Splice Sta 368- 434 | 53-80249-01 | This EO complies with 2016-20-03 par (g) Federal Aviation Administration (FAA) Airworthiness Directive AD 2003-18-10 is related to this service bulletin. The effective date of AD 2003-18-10 is October 16, 2003. This service bulletin affects SSI 53-10-107A of MPD 622T001-9 Section 9, which is mandated by AD 2003-18-10. All other MPD inspections outside of the affected area, including SSI 53-10-107B, are not affected and must continue. Boeing recommends that the inspection and repair given in this service bulletin be done in accordance with Paragraph 1.E., Compliance, Table 1. Do the required actions in accordance with Paragraph 3. Accomplishment Instructions. |
| 2003-19-09 | 767-51A0023 | Inspect - Canted Pressure Deck Drain Lines & Clean Canted Pressure Bulkhead Cavity | 53-80106-02 | This EO complies with the repetitive actions (WORK PACKAGES 1, 2, AND 3) of S/B 767-51A0023, Revision 1 dated 8/27/2009, and paragraphs (a)(b)(c) and (d) of AD 2003-19-09, amendment 39-13312. |
| 2003-23-05 | 767-26A0121 | Inspect - Hoses, Cargo Compartment Fire Extinguisher | 26-80140-03 | Compliance with this EO will accomplish the modification/repair specified in SB 767-26A0121, R1, dated 2/2/2006 as mandated by AD 2003-23-05, AMND. 39-13369, effective 12/24/2003. |
| 2003-26-14 (e), 2010-01-03 (g) | Kidde Aerospace S/B 898052–26–449 | Inspect - Hand Held Halon Fire Extinguishers | 26-80230-02 | This task complies verifies compliance with the following ADs: Paragraph (e) of AD 2003-26-14, amendment 39-13413, effective date 02/20/2004 Paragraph (g) of AD 2010-01-03, amendment 39-16159, effective date 01/20/2010. |
| 2004-05-07 | 767-24A0134 | Inspect - Wire Bundles Below P37 Panel | 24-80172-02 | Accomplishment of this task complies with Boeing Service Bulletin 767-24A0134, Rev 01, dated 10/18/2001 as mandated by AD 2004-05-07, amendment 39-13501, effective 04/07/2004 paragraphs (a) and (b). |
| 2004-10-05 | 767-34A0332 | Inspect/T est - ADC Switching System | 34-80235-01 | This EO inspects/tests the installation of ADC Switching System for the purpose of verifying compliance with Boeing Service Bulletin 767-34A0332, Rev. 05 dated 12/18/2008 as required by paragraph (c)(3) of AD 2004-10-05, amendment 39-13635 effective date 06/22/2004. Boeing Service Bulletin 767-34A0332, Rev. 05 dated 12/18/2008 has been approved as an alternative method of compliance (AMOC) to the requirements of paragraph (c) of AD 2004-10-05. |
| 2004-16-11 | 767-29-0103 Parker SB 6513902- 29-305 | Inspect - Ram Air Turbine (RAT) Hydraulic Pump Serial Number | 29-80171-02 | This EO accomplishes Boeing Service Bulletin 767-29-0103, dated 09/12/2002 as mandated by AD 2004- 16-11, AMND 39-13767, Dated 09/13/2004 |
| 2004-16-12 | 767-29-0057 767-29A0115 | INSP/MOD-ACMP WIRE & BRACKET, P&W 4000 | EA 60658 | THIS EO ACCOMPLISHES THE INTENT OF SB 767-29A0115, R0, DTD 5/22/2013 AND IS AN AMOC FOR THE REQUIREMENTS OF PARAGRAPH (B) OF AD·2004-16-12 THAT REQUIRES INCORPORATION OF SB 767-29-0057 WHICH INSTALLS THE BRACKETS THAT ARE REPLACED BY THIS EO. |
| 2005-02-02 | 767-53A0113 | Inspect - DVI & HFEC, Nose Wheel Well Vertical Chords, STA 287, LBL 25 & RBL 25 | 53-80107-05 | Compliance with this EO accomplishes the repetitive inspection of S/B 767-53A0113 R0, dated 02-26-04 as mandated by paragraph (h) of AD 2005-02-02, amdt 39-13947, effective 03-01-05. |
| 2005-05-20 | 767-52-0087 | Verify- Flight Compartment Door Decompression Sensing Module Change | 52-80183-01 | This EO verifies that Boeing SB 767-52-0087 Rev. 1 Dated October 12, 2006 has been accomplished. If so, this satisfies the requirements of AD 2005-05-20 Amendment 39-14555, Effective April 17, 2006. |

| AD | S/B | Subject | EO | Statement of compliance |
|------------|---|---|-------------|---|
| 2005-11-02 | 767-53A0078 | Inspect - STA 1809.5 Permanent Repair | 53-80141-04 | NOTE: AD 2005-11-02 superseded AD 2001-09-13, effective July 1, 2005, Amendment 39-12220. This EO accomplishes the intent of AD 2005-11-02, amendment 39-14104, dated 07/01/2005, para. (L)(1)(i) for a/c that do not have the permanent repair installed, AND, para (n)(1) for a/c that DO have the permanent repair installed. Prior to permanent repair: Job Card 53-80141-04-101 complies with AD 2005-11-02 para. (L)(1)(i) Job Card 53-80141-04-201 complies with AD 2005-11-02 para. (L)(1)(i) Post permanent repair: Job Card 53-80141-04-102 complies with AD 2005-11-02 para. (n)(1) Job Card 53-80141-04-102 complies with AD 2005-11-02 para. (n)(1) Post permanent repair for a/c MSN 22215 only: Job Card 53-80141-04-302 complies with AD 2005-11-02 para. (n)(1), and 8110-3 AMOC Aug 1, 2002, reinspection requirements for 26 holes oversized, with oversized fasteners installed. Right side only. |
| 2005-17-09 | 767-25-0290 | Inspect/Verify - Flight Compartment, Drip Shield, Fire Block Installation | 25-80233-01 | This EO along with previous operator compliance records will verify that S/B 767-25-0290, rev. 5, has been accomplished as mandated by AD 2005-17-09, amendment 39-14230 effective date 09/27/2005. S/B 767-25-0290, rev. 5 dated 06/08/2006 has been approved as an AMOC to the requirements of AD 2005-17-09. |
| 2006-07-14 | 767-32A0199 767-32A0202 | Inspect - Bogie Beam Pivot Pins | 32-80222-01 | This task complies with the intent of Boeing S/B's 767-32-0199 R02, Dated 04/08/2004 and 767-32A0202, R01, Dated 06/09/2005 as mandated by AD 2006-07-14, Amendment 39-14541, effective 05/12/2006 paragraphs (g) and (h). |
| 2006-08-04 | 767-28A0071 | Inspect/Rework - Engine Fuel Feed Tube Fitting, Electrical Bond | 28-80209-01 | Compliance with this EO accomplishes the requirements of Boeing Service Bulletin 767-28A0071, R2 dated 11/08/2005 as mandated by paragraph (g) of AD 2006-08-04, amendment 30-14558, effective 05/18/2006. |
| 2006-09-09 | 767-53-0118 | Inspect - DVI/HFEC, Skin, Bulkhead Outer Chord and Strap, Right STA 1725.5 | 53-80248-01 | Accomplishment of this EO complies with the intent of Boeing Service Bulletin 767-53-0118 R01 dated January 12, 2012 as mandated by paragraph (g) of AD 2006-09-09, amendment 39-14580 dated June 7, 2006, S/B 767-53-0118, R01 has been approved as an AMOC to the requirements of AD 2006-09-09 paragraphs (f), (g) and (h). See terminating action in engineering notes. |
| 2006-10-08 | 767-27A0167 | Inspect - Trailing Edge Flap System - Flap Mechanisms 3-10 Links | 27-80166-02 | This EO complies with Part 3 of Boeing S/B 767-27A0167 Rev 3, dated 12/19/2007, for Group 1 airplanes, and AD 2006-10-08 amendment 39-14592 dated 06/19/2006, paragraph (i)(1) or (j). SB 767-27A0167, Rev 3 has been approved as an AMOC to the requirements of AD 2006-10-08. |
| 2006-26-13 | 767-25-0288 | Inspect - Shoulder Restraint Harness C-clips | 25-80142-02 | This EO accomplishes the intent of service bulletin 767-25-0288, revision 3, as mandated by AD 2006-26-13, effective 02/12/2007. |
| 2007-10-03 | 767-26A0123 767-26A0130 767-26-0118 | Inspect - Fire Extinguishing System, Aft Cargo Compartment | 26-80143-02 | Compliance with this EO will accomplish the intent of S/B 767-26A0130, Rev 4 dated 01/20/2011 as mandated by AD 2007-10-03, par. (f)(1) and S/B 767-26A0123, dated 08/22/2002 as mandated by AD 2007-10-03, par. (f)(2)((i). If discrepancies are found concerning insufficient clearances called out in S/B 767-26A-123, modifications IAW S/B 767-26-0118, Rev 5, dated 09/16/2011 will be accomplished as mandated by AD 2007-10-03, para (f)(2)(ii). S/B 767-26A0123, Rev 4, dated 01/20/2011 has been approved as an alternative method of compliance to the requirements of paragraph (f)(1) of AD 2007-10-03. S/B 767-26A0118, Rev 5, dated 09/15/2011 has been approved as an alternative method of compliance to the requirements of paragraph (f)(2)(ii) of AD 2007-10-03. |

| AD | S/B | Subject | EO | Statement of compliance |
|----------------|---|---|-------------|--|
| 2007-16-18 | 767-25-0376 | Inspect / Clean - FWD & AFT Lower Lobe Wiring Bundles | 25-80109-02 | This EO accomplishes the inspection/cleaning of Service Bulletin 767-25-0376, Revision 2, as mandated by AD 2007- 16-18, paragraph (f), effective 09/25/2007. S/B 767-25-0376, Rev. 2 has been approved as an AMOC to the requirements of paragraph (f) of AD 2007-16-18. (f) Within 36 months after the effective date of this AD, do detailed inspections for damage to the wire bundles, power drive unit wiring, and wire attaching hardware, supports, and sleeving located in the forward and aft lower lobe cargo compartments; and do all applicable corrective actions before further flight after the inspections; by accomplishing all of the actions specified in the Accomplishment Instructions of Boeing Service Bulletin 767-25-0376, Revision 1, dated February 9, 2007 (for Model 767-200, -300, and -300F series airplanes); or Boeing Service Bulletin 767-25-0377, Revision 1, dated February 9, 2007 (for Model 767-400ER series airplanes); as applicable. Repeat the inspections thereafter at intervals not to exceed 24,000 flight hours or 72 months, whichever occurs first. |
| 2007-23-09 (f) | 767-28-0087 Cinch Service Bulletin CN1156-28- 02 | Inspect - Fuel Quantity Indication System, Tank Wire Harness Connectors | 28-80231-01 | This task complies with the intent of Boeing S/B 767-28-0087, Rev 03, dated 01/12/2012 as mandated by paragraphs (f)(1) and (f)(2) of AD 2007-23-09, amendment 39-15255, effective 12/20/2007. S/B 767-28-0087, Rev. 03 has been approved as an AMOC to the requirements of AD 2007-23-09, paragraph (f). |
| 2007-24-08 | 767-27-0197 | Inspect - Rudder and Elevator, Freeplay and Lubrication | 27-80110-05 | Completion of EO-27-80110-XX complies with the intent of Boeing S/B 767-27-0197 R1 dated 07/19/2007 as mandated by paragraphs (g), (h), and (j) of AD 2007-24-08, amendment 39-15274 dated 11/28/2007. Accomplishing paragraphs (j), (k) and (l) of AD 2013-25-03 also complies with paragraph (k) of AD 2007-24-08 as is stated in paragraph (o) of AD 2013-25-03. JMCC EO-27-80121-XX, Job Cards 27-80121-XX-201 & -301 comply with paragraphs (j), (k)(1) and (l) of AD 2013-25- 03 and EO-27-80126-XX, job cards 27-80126-XX-101 and 27-80126-XX-201 comply with paragraph (k)(2) of AD 2013 25-03. |
| 2008-06-06 | 767-27A0194 | Inspection - Detailed and Freeplay with Lubrication - Horizontal Stabilizer Actuator | 27-80111-02 | This EO contains three maintenance tasks split between three job cards. Each task has a different repetitive interval for compliance. Only the job cards for the maintenance tasks due will be issued with this EO. See page 5 for the issued job cards and note the compliance for each below. JC 27-80111-02-001 -Inspection - Detailed, Horizontal Stabilizer Actuator. This card accomplishes Work Package 1 of Boeing SB 767-27A194 Rev. 3, Dated November 13, 2008 (Approved AMOC) and satisfies only that portion of par (f) of AD 2008-06-06, amendment 39- 15418 effective April 16, 2008. JC 27-80111-02-002 -Lubrication - Horizontal Stabilizer Actuator, Ballnut and Ballscrew. This card accomplishes Work Package 2 of Boeing SB 767-27A194 Rev. 3, Dated November 13, 2008 (Approved AMOC) and satisfies only that portion of par (f) of AD 2008-06-06, amendment 39-15418 effective April 16, 2008. JC 27-80111-02-003 -Inspection - Stabilizer Trim Ballscrew & Ballnut Freeplay. This card accomplishes Work Package 3 of Boeing SB 767-27A194 Rev. 3, Dated November 13, 2008 (Approved AMOC) and satisfies only that portion of par (f) of AD 2008-06-06, amendment 39-15418 effective April 16, 2008. JC 27-80111-02-003 -Inspection - Stabilizer Trim Ballscrew & Ballnut Freeplay. This card accomplishes Work Package 3 of Boeing SB 767-27A194 Rev. 3, Dated November 13, 2008 (Approved AMOC) and satisfies only that portion of par (f) of AD 2008-06-06, amendment 39-15418 effective April 16, 2008. |
| 2009-06-08 | 767-57A0097 | Inspect - Wing Skin - AFT Pitch Load Fitting | 57-80167-03 | Compliance with this EO accomplishes the repetitive inspection requirements and modification/repair as described in SB 767-57A0097, R1, dated 10/18/2007 as mandated by AD 2009-06-08, amendment 39-15844, dated 04/28/2009. SB 767-57A0097, R2 dated 11/10/2014 has been issued and incorporates data contained in 767- 57A0097 IN-01, IN-02, AMOC-01 and AMOC-02. SB 767-57A0097, R2 has been approved as an AMOC to AD 2009-06 08, Para. (f), (g), (h), (i), (j) and (k). Corrections contained in SB 767-57A0097, R2 have been incorporated into this EO where applicable. |

| AD | S/B | Subject | EO | Statement of compliance |
|------------|---|--|-------------|---|
| 2009-16-06 | 767-28-A0083 | Inspect Auto Shutoff System, Center Tank Pumps | 28-80255-01 | This EO verifies compliance with paragraph (f) of AD 2009-16-06, amendment 39-15989 dated 10/09/2009. Paragraph (g) states if installation of automatic shutoff system was accomplished before the effective date of the AD it is acceptable to take compliance with paragraph (f) using S/B 767-28A0083, original issue or revision 1. Accomplishment of AD 2009-16-06 terminates the requirements of AD 2011-25-05 paragraph (j). |
| 2009-21-07 | Middle River Aircraft Systems CF6-80C2 S/B 78A1162 | Inspect/Replace - No1 & No 2 - Fan Reverser - Adjustable-Length Actuator - Rod Ends | 78-80173-03 | Accomplishment of this EO complies with Middle River Aircraft Systems CF6-80C2, S/B 78A1162, R3 dated 9-10-09, as mandated by AD 2009-21-07, amendment 39-16044 dated 12-01-09. Job cards 78-80173-03-101 thru -104 perform repetitive replacement of adjustable length actuator, 3/8-inch rod ends and complies with AD 2009-21-07 paragraph (g)&(h). Job cards 78-80173-03-201 thru -204 perform inspection of Clevis attaching hardware, and complies with AD 2009- 212-07 paragraph (f). Job cards 78-80173-03-301 thru -304 perform terminating action by replacing adjustable length actuator, 3/8-inch rod ends with 7/16-inch rod ends or replacing the adjustable length actuator with a fixed end actuator and complies with AD 2009-21-07 paragraph (i). |
| 2009-23-04 | 767-28A0095 | Inspect/Replace - Main Tank, Fuel Boost Pump | 28-80112-03 | This EO verifies/accomplishes the requirements of Boeing S/B 767-28A0095 as mandated by AD 2009-23-04, amendment 39-16076, effective 12/14/2009 paragraph (g) and (h). The EO verifies/accomplishes both initial and repeat replacement of the fuel boost pumps IAW the AD intervals. |
| 2010-03-08 | 767-54A0062 | Inspect/Replace - Midspar Fitting Fuse Pins | 54-80188-02 | This EO accomplishes the repetitive inspection of the Midspar Fuse Pins I/A/W SB 767-54A0062, Rev. 07, dated 09/09/10 mandated by AD 2010-03-08, Amendment 39-16192, dated 02/19/2010. SB 767- 54A0062, Rev. 07 has been approved as an AMOC to the requirements of AD 2010-03-08 par (g), (h), (i), (n), (o) and (p). AD 2010-03-08 supersedes AD 2003-03-02 which superseded AD 94-11-02. |
| 2010-06-10 | 767-28A0094 | Inspect/Verify - Fuel Qty. Indicating System, Hot Short Protection | 28-80229-01 | This task complies with the intent of Boeing S/B 767-28A0094, Rev. 03, dated 09/19/2011 as mandated by AD 2010-06-10, amendment 39-16234, effective 05/03/2010 paragraph (f) (1). S/B 767-28A0094, Rev. 03, dated 09/19/2011 has been approved as an AMOC to the requirements of AD 2010-06-10 paragraph (f). |
| 2010-14-18 | 767-54A0101 | Inspect - Open Hole, HFEC Pylon Midspar Fitting | 54-80187-02 | Compliance with this EO incorporates part 2 of SB 767-54A0101, Rev 5, dated 06/29/2010, as mandated by para (g)(2) of AD 2010-14-18, Amnd. 39-16363, dated 08/06/2010. AMOC's 767-54A0101-AMOC-01 and 767-54A0101-AMOC-02 have been incorporated in this EO to ensure correct surface finish and corrosion protection. |
| 2010-15-01 | 767-30-0039 | Inspect - Left and Right Windshield T erminal, Connector Lug | 30-80132-04 | This EO complies with the accomplishment instructions for the repetitive inspections contained in Boeing S/B 767- 30-0039, Rev, 1 dated Dec. 16, 2011 as required by AD 2010-15-01, amendment 39-16367 dated Aug. 17, 2010. L1 or R1 Windshield replacement, resulting from inspection findings, using Top Kit P/N 015T0500-1 or 015T0500-2 respectively, will constitute terminating action for repetitive inspections IAW AD 2010-15-01 paragraph (i). |
| 2010-22-01 | 767-54A0074 | Inspect/Replace - Pylon, Upper Link Fuse Pins | 54-80226-02 | Compliance with this EO accomplishes SB 767 54A0074, Rev 01, dated 04/24/2008 as mandated by AD paragraph (g) or (j) of 2010-22-01, amendment 39-16480, dated 11/04/2010. 767-54A0074-AMOC-01, dated 12/28/2011 and 767-54A0074 IN03 have been incorporated in this EO. Not |
| 2010-23-03 | 767-28-0106 | FUEL - Fuel Quantity Indicating System-Low Fuel Quantity Indication Test | 28-80113-02 | This EO complies with Boeing SB 767-28-0106, Revision 1, dated December 22, 2010, Part 1 for Group 1 airplanes, as mandated by AD 2010-23-03, Amendment 39-16492, Paragraphs (g), (h), & (i)(2). Boeing SB 767-28-0106, Revision 1 is approved as an AMOC to AD 2010-23-03, Amendment 39-16492, Paragraphs (g), (h), & (i)(2). |

| AD | S/B | Subject | EO | Statement of compliance |
|---|--|---|-------------|---|
| 2010-26-08 | 767-25-0477 | Inspect / Repair - Wiring above Entryway Doors | 25-80114-02 | IAI has received an AMOC to AD 2010-26-08 (ref. FAA letter 150s-11-68, dated Feb 23, 2011) for compliance with the inspections required by paragraphs (g)(1), (g)(2) and (g)(3). The AMOC does not apply to the inspection requirements of paragraph (g)(4) which is a detailed inspection of the wiring specified in SB 767-25-0477. This EO accomplishes the intent of paragraph (g)(4). SB 767-25-0477, R2 has been approved as an AMOC to the requirements of paragraphs (g)(1), (g)(2), (g)(3), and (g)(4) of AD 2010-26-08. |
| 2011-01-09, 2016-11-20, 2017-18-12, | B/E Aerospace Service Bulletin No. 119003-35-009 B/E Aerospace Service Bulletin No. 119003-35-011 B/E Aerospace Service Bulletin No. 119003-35-013 B/E Aerospace Service Bulletin No. 119003-35-5 | Inspect/Replace - B/E Aerospace Protective Breathing Equipment (PBE) | 35-80185-03 | This EO complies with B/E Aerospace Service bulletins; 119003-35-009, rev 001 dated April 12, 2016, 119003-35- 011, rev 000, dated Feb 04, 2015 and 119003-21, rev 002 dated 07/19/2017 as mandated by paragraphs (h), (i) and (j) of AD 2017-18-12, amendment 39-19021, dated 10/16/2017. Compliance with AD 2017-18-12 paragraph (h)(2) also complies with AD 2011-01-09. |
| 2011-02-06 | 767-53-0207 | Inspect - Lower VHF Antenna, External Skin | 53-80144-02 | Compliance with this EO accomplishes the intent of Boeing SB 767-53-0207, dated 12/17/09, as mandated by AD 2011-02-06, paragraph (g), amendment 39-16579, effective 03/01/11. |
| 2011-14-02 | | Inspect - Section 48 - Main Frame and Skins - STA 1809.5 Bulkhead - Fwd Outer Chord, and Horizontal & Vertical Inner Chord | 53-80164-04 | This EO accomplishes the repetitive inspections and repairs IAW S/B 767-53A0131, Rev. 02 dated 09/18/2012 as mandated by AD 2011-14-02, paragraphs (g), (h) and (k). S/B 767-53A013, Rev. 02 has been approved as an AMOC to the requirements of paragraph (g) and (k) of AD 2011-14-02. If while accomplishing S/B Part 1 and Part 2, cracks are found in the skin or in any structure other than the forward outer chord, horizontal inner chord, or vertical inner chord during any inspection required by paragraph (g) or (k) repairs must be made before further flight as mandated by paragraph (h) of AD 2011-14-02. |
| 2011-25-05 | 767-28-A0085 TDG Document 75- 3206-001 | Inspect - Fuel Pump Control Relays, Main Tank (Wing) | 28-80253-01 | This EO verifies compliance with paragraphs (g) and (h)(1) of AD 2011-25-05, amendment 39-16881 dated February 22, 2012. Installation of approved GFI relays IAW TDG Aerospace document 75-3206-0001. TDG Aerospace document 75-3206-0001 has been approved as an AMOC to the requirements of AD 2011-25-05 paragraphs (g) and (h)(1). |
| 2011-25-11 | 767-31-0098 767-31-114 767-31-267 767-31-295 | EICAS Computer and Software Upgrade | 31-80161-04 | Compliance with this EO will verify the accomplishment of the intent of AD 96-07-09, amendment 39-9558, dated 05/03/1996, paragraph (d)(1), (EICAS computer upgrade) and the intent of AD 2011-25-11, amendment 39-16887, dated 01/17/2012. paragraph (i)(2), (OPS Version 7 upgrade) and (j)(8)&(12), (EICAS computer and OPC upgrade). Compliance with this EO will also verify the concurrent requirement of AD 2004-10-05, amendment 39- 13635, dated 06/22/2004 paragraph (d)(2). SB 767-31-295, IR, dated 06/16/2014 further upgrades the Operational Program Software (OPS) to Version 8. SB 767-31-295 has been approved as an AMOC as an alternative method of compliance to the requirements of paragraph (d)(1) of AD 96-07-09 and as an alternative method of compliance to the requirements of paragraph (i)(2) of AD 2011-25-11 |
| 2012-02-08 | ACSS Service Bulletin 7517900- 34-6048 | Inspect - ACSS TCAS Computer Installation | 34-80170-02 | Completion of this EO will verify the manufacturer of the TCAS Computer to determine the status of AD 2012-02- 08, amendment 39-16931, dated 03/13/2012. |

| AD | S/B | Subject | EO | Statement of compliance |
|------------|----------------------------|---|-------------|--|
| 2012-05-04 | 767-56A0010 767-56-0004 | INSP/MOD - #2 WINDOWS, FLIGHT COMP ARTMENT | 56-80115-05 | Compliance with this EO accomplishes the inspection and/or re-rigging of the link arm of the left and right #2 openable window in the flight compartment. If satisfactory re-rigging is not possible, modification of the window link arm mechanism shall be accomplished IAW S/B 767-56A0010 Rev. 3 dated March 3, 2011 as mandated by AD 2012-05-04, Amendment 39-16976 dated April 24, 2012. The S/B initial inspection is not required if the S/B modification is accomplished. See T able 1 of paragraph 1.E. of the S/B. |
| 2012-08-14 | 767-57A0117 | Inspect/Repair - Upper Wing Skin Holes at Front Spar Pitch Load Fitting | 57-80224-01 | This EO complies with the intent of Boeing S/B 767-57A0117, R2, dated 01/22/2013 as mandated by paragraph (g) and (h) of AD 2012-08-14, amendment 39-17031, effective 05/31/2012. |
| 2012-09-04 | 767-53A0100 | Inspect - Rear Spar Bulkhead Side Fitting, Failsafe Straps at Station 955 | 53-80116-03 | This EO complies with the repetitive detailed inspection, eddy current and ultrasonic inspection of SB 767- 53A0100, Rev. 3, dated 02/06/2012 as mandated by paragraphs (i) and (j) of AD 2012-09-04 dated 06/11/2012 |
| 2012-11-09 | | Inspect/Remove - Lavatory Chemical O2 Generators | 35-80207-01 | This task verifies compliance with AD 2012-11-09 R1, amendment 39-18221, dated 07/27/2015 paragraphs (g) and (l). |
| 2012-12-14 | 767-53A0228 | Inspect/Repair - DVI/HFEC, Lower Main Sill Inner Chord, STA 883.5 | 53-80263-01 | This EO complies with inspections and if necessary repairs specified in S/B 767-53A0228 dated 07/28/2011 as mandated by paragraph (g) of AD 2012-12-14, amendment 39-17094 dated 07/30/2012. |
| 2012-15-12 | 767-57A0126 | Inspect/Replace - Lower Underwing Longeron Fitting at FWD Intersection of Fuselage and Wing Skin | 57-80147-02 | Completion of JMCC EO-57-80147-01 complies with the requirements of AD 2012-15-12, Amendment 39- 117141, paragraph (g). Service Bulletin 767-57A0126, Revision 4 is approved as an AMOC to paragraph (g) of the AD. |
| 2012-21-02 | 767-57-0121 | Modify - Leading Edge Fluid Drain Path | 57-80117-02 | Compliance with this EO accomplishes the modification/repair of SB 767-57-0121 R3, dated 12/03/2012 (including changes incorporated by 767-57-0121 AMOC 01 dated 05/28/2014) as mandated by AD 2012-21- 02 amnd. 39-17218, dated 11/27/2012. S/B 767-57-0121, R3 has been approved as an AMOC to the requirements of AD 2012-21-02 para (g). |
| 2012-22-17 | 767-53A0209 | Inspect/Repair - ST A 859.5, 883.5, and 903.5 Inner Chords | 53-80163-02 | Accomplishment of this EO complies with the intent of Boeing S/B 767-53A0209, R02 as mandated by AD 2012-22- 17, amendment 39-17255, dated 12/26/2012, paragraph (g). S/B 767-53A0209, Revision 2, has been approved as an Alternative Method of Compliance (AMOC) to the inspection and repair actions required by paragraph (g) of AD 2012-22-17. |
| 2013-05-07 | 767-51A0027 | Airframe Drainage - Gutter Installation, Horizontal Stabilizer Center Section | 51-80148-03 | This EO complies with, or verifies previous compliance with, SB 767-51A0027, R1 dated October 12, 2006 as mandated by the requirements of AD 2013-05-07, par. (j) and (k). |
| 2013-11-04 | 767-25-0381 | Inspect/Replace - BMS8-39 Foam Insulation Seals | 25-80225-02 | Compliance with this EO accomplishes the inspections/modifications of SB 767-25-0381, Rev. 3, dated 11/02/2015. Pargraph (g) of AD 2013-11-04, Amdt. 39-17464, dated 07/09/2013, mandates compliance IAW S/B 767-25-0381, Rev 1, dated 09/17/2012. S/B 767-25-0381, Rev. 3 has been approved as an AMOC to the requirements of paragraph (g) of AD 2013-11-04. |
| 2013-13-08 | 767-57A0102 | Inspect / Install - Fasteners for Wing Fuel Tanks | 57-80149-02 | This EO provides instruction to change wire bundle clamp configuration at various locations on the front spar and right wheel well as well as seal fasteners & stiffeners in the fuel tanks work packs 1 thru 7 of SB 767- 57A0102, R5, Dated 04/01/2013 and paragraph (h) of AD 2013-13-08 Amdt 39-17496 Dated 8/20/13. This EO provides instruction to inspect wire bundle clamp configuration at various locations in the front spar and right wheel well IAW workpacks 13-15 of SB 767-57A0102, Revision 5, Dated 4/1/2013 and paragraph (k) of AD 2013-13-08 Amdt 39-17496 Dated 8/20/13 |
| 2013-18-02 | 767-57A0131 | Inspect/Replace - Flap Support Rib Nuts/Bolts | 57-80119-02 | This EO complies with the terminating action (part 3) of SB 767-57A0131 R01, dated 05/08/2013, and paragraph (h)(1)(ii) of AD 2013-18-02, amendment 39-17575, effective 10/22/2013, on each flap support rib location. |
| 2013-24-10 | 767-24-0200 | Modify - Battery Charger System | 24-80151-01 | Compliance with this EA accomplishes the modification of the battery charger system IAW S/B B767-24-0200 R2, dated 03-26-14, as mandated by AD 2013-24-10, amendment 39-17684, effective 01-07-2014. SB B767-24-0200 R2 has been approved as an AMOC to the requirements of paragraph (g) of AD 2013-24-10. |

| AD | S/B | Subject | EO | Statement of compliance |
|------------|----------------------------|--|-------------|--|
| 2013-25-02 | 767-28A0053 767-28A0104 | Inspect/Replace - Engine Fuel Boost and Override/Jettison Pump Wire Bundles | 28-80168-05 | Job card 28-80168-XX-110 and -120 accomplishes the repetitive inspections called out in Boeing Service Bulletin 767-28A0053, Revision 3, dated November 11, 2011 as mandated by paragraphs (g), (h), (i), (j) & (k) of AD 2013-25- 02, dated June 5, 2014. Job Card 28-80168-XX-210, -211, -220 -230, -310 and -320 accomplishes terminating action for paragraphs (g) and (k) of AD 2013-25-02 by replacing the fuel boost pump and the override/jettison pump wire bundles inside the in- tank electrical conduit IAW Boeing Service Bulletin 767-28A0104, Revision 2, dated November 28, 2016 as mandated by paragraphs (l) of AD 2013-25-02, dated June 5, 2014. AD 2013-25-02 paragraph (j) is complied with by JC-28-80168-XX-230 following wire bundle replacement. Boeing Service Bulletin 767-28A0104, Revision 2, dated November 28, 2016 has been approved as an AMOC to the requirements of paragraph (l) of AD 2013-25-02. |
| 2013-25-03 | 767-27A0168 767-27-0202 | Inspect/Replace - Elevator Power Control Actuator (PCA) Bellcrank | 27-80121-05 | Job Card 27-80121-05-101 - Accomplishes the intent of S/B 767-27A0168, Rev. 01, dated 06/16/2011 as mandated by paragraph (h) of AD 2013-25-03, amendment 39-17699 effective 03/03/2014. Paragraph (h) is a restatement of superceded AD 2001-04-09 paragraph (a)). S/B 767-27A0168, Rev. 01 has been approved as an AMOC to the requirements of paragraph (a) and (b) of AD 2001-04-09. Job Cards 27-80121-05-201 & -301 - Accomplishes the intent of S/B 767-27-0186, dated 06/25/2007 as mandated by paragraph (j), (k)(1) and (l) of AD 2013-25-03, amendment 39-17699 effective 03/03/2014. Complying with Paragraphs (j), (k)(1) and (l) of AD 2013-25-03 terminates the requirements of paragraphs (g) and (h) of AD 2013-25-03. EO 27-80121-05, Job Cards 27-80121-05-201 & -301 comply with paragraphs (j), (k)(1) and (l) and EO-27-80126-03, job cards 27-80126-03-101 and 27-80126-03-201 comply with paragraph (k)(2). Complying with both of these EOs/job cards terminates the requirements of paragraphs (g) and (h) of AD 2013-25-03. |
| 2013-25-03 | 767-27-0200 767-27-0202 | Functional Check - Elevator PCA Input Pogo/Rod and Mis-Rig Check | 27-80126-04 | Job cards 27-80126-XX-101 and -201 accomplishes the repetitive (12,000 FH) functional test (pogo check) on each of the six elevator PCA input rod assemblies, in accordance with the Accomplishment Instructions of SB 767-27- 0200, dated 06/25/2007, as mandated by paragraph (k)(2) of AD 2013-25-03, R00, amendment 39-17699 dated 03/03/2014. This functional test also satisfies MPD Item 27-078-00. Job cards 27-80126-XX-102 and -202 accomplishes the repetitive (6,000 FH) check of the elevator PCA rigging, in accordance with the Accomplishment Instructions of Boeing Service Bulletin 767-27-0202, Revision 1, dated February 21, 2008 as mandated by paragraph (I)(2) of AD 2013-25-03, R00, amendment 39-17699 dated 03/03/2014. This test also satisfies MPD Item 27-079-00. NOTE: The tasks in the EO should only be performed after previous accomplishment of Paragraphs (j), (k) and (l) of AD 2013-25-03. Complying with Paragraphs (j), (k) and (l) of AD 2013-25-03 terminates the requirements of paragraphs (g) and (h) of AD 2013-25-03. EO 27-80121-XX, Job Cards 27-80121-XX-201 & -301 comply with paragraphs (j), (k)(1) and (l)(1). |

| AD | S/B | Subject | EO | Statement of compliance |
|------------|--|---|-------------|--|
| 2014-05-16 | 767-78A0081 767-78A0100 MRAS S/B 78-1168 | Replace/Test - Thrust Reverser Control System, Center Drive Unit (CDU) | 78-80169-02 | This EO complies with the following: Job cards -101, -102, -103 and -104 accomplish replacement of the short flexshaft on each thrust reverser half IAW S/B 767-78A0100, dated 10/26/2010 as mandated by paragraph (g) of AD 2014-05-16, amendment 39-17789 dated 04/25/2014. Job Cards -201, -202, -203 and -204 accomplish the repetitive repetitive functional tests of the Thrust Reverser, Electro-Mechanical brakes and the Center Drive Unit (CDU) cone brakes IAW S/B 767-78A0081, Rev 02 dated April 19, 2001 as mandated by paragraph (h) of AD 2014-05-16, amendment 39-17789 dated 4/25/2014. Any failures noted during functional test shall be investigated and corrected before further flight as mandated by AD 2014-05-16, para (i). |
| 2014-05-27 | Rockwell Collins Service Information Letter 13-1 | Inspect - ATC Mode S Transponder | 34-80152-01 | This EO verifies compliance with or complies with the requirements of paragraph (g) of AD 2014-05-27, amendment 39-17799, dated 05/02/2014. |
| 2014-14-04 | 767-53A0260 | INSPECT/REPAIR LAP SPLICE STA 368 - 434 | NEA61884 | COMPLIANCE WITH THIS EO INCORPORATES SB 767-53A0260, REVISION 00 DATED 08/26/2014. AD 2014-14-04, AMENDMENT 39-17899 DATED 9/5/2014, WHICH SUPERSEDES AD 2003-18-10, AMENDMENT 39-13301 DATED 9/11/2003 REQUIRES REVISION OF THE 767 MAINTENANCE PROGRAM TO INCORPORATE THE LIMITATIONS SECTION IN SUBSECTION B, AIRWORTHINESS LIMITATIONS-STRUCTURAL INSPECTIONS, OF SECTION 9, AIRWORTHINESS LIMITATIONS (AWLS) AND CERTIFICATION MAINTENANCE REQUIREMENTS (CMRS), D622T001-9, REVISION JULY 2011 OR REVISION FEBRUARY 2014, OF THE BOEING 767 MPD DOCUMENT. THIS EO AND SB 767- 53A0260, ORIGINAL ISSUE ARE RELATED TO D622T001-9 STRUCTURAL SIGNIFICANT ITEM 53-10-107A. |
| 2014-17-02 | Honeywell Alert Service Bulletin 1152682-23-A22 | Inspect - Honeywell ELT | 25-80240-01 | This EO verifies compliance or complies with Honeywell Alert Service Bulletin 1152682-23-A22, Revision 1, dated 08/08/2013 as required by AD 2014-17-02, amendment 39-17955, effective date 09/09/2014 all paragraphs. |
| 2014-20-02 | 767-28-0105 | Inspect - Fuel Tank Access Door | 28-80242-01 | Accomplishment of this EO complies with the intent of S/B 767-28-0105, Rev 01 dated 02/06/2013 as mandated by paragraph (g) (1) and (2) of AD 2014-20-02, amendment 39-17975 dated 11/05/2014. |
| 2014-22-09 | 767-27A0229 | Inspection - Flap Rotary Actuator and Reaction Ring | 27-80139-03 | This EO provides instructions to comply with the repetitive detailed inspections and corrective actions, if necessary, of SB 767-27A0229, Revision 01, Dated June 03, 2016, and paragraph (g) of AD 2014-22-09, Amendment 39-18014 at TE Flap - support positions 1, 2, 3, 4, 5, 6, 7, and 8. Service bulletin 767-27A0229, R1 dated 06/03/2016 has been approved an alternative method of compliance to the corresponding inspections and related investigative and corrective actions specified in paragraph (g) of AD 2014-22-09. |
| 2015-05-01 | 767-29A0110 767-29-0112 Hamilton Sundstrand SB 729548-29-15 Hamilton Sundstrand SB 729548-29-18 | Inspect - Ram Air Turbine Hub Assembly and Balance Washer Retention Screws | 29-80154-03 | Compliance with this EO accomplishes the intent of SB 767-29A0110, Rev 1, Dated 03/08/2010 as mandated by AD 2015-05-01 amendment 39-18111, dated 04/20/2015, paragraph (g) as well as the intent of S/B 767-29A0112, dated 06/24/2010 as mandated by AD 2015-05-01 amendment 39-18111, dated 04/20/2015, paragraph (h). |
| 2015-08-07 | Zodiac Aerospace SB MHX-35-240 | Inspect/Replace - Zodiac Oxygen Regulator Inflatable Harnesses | 35-80122-03 | This EO complies with the accomplishment instructions of Zodiac Aerospace SB MHX-35-240, Rev. 07, Dated 09/01/2011 as mandated by AD 2015-08-07, Amendment 39-18143, dated June 16, 2015. |

| AD | S/B | Subject | EO | Statement of compliance |
|------------|---|---|-------------|--|
| 2015-12-11 | 767-52A0100 | Inspect/Replace-Lower Lobe Large Cargo Door, Rotary Actuators | 52-80123-02 | This EA address and complies with the intent of Boeing S/B 767-57A100, R03 and Eaton S/B 692D100-52- 4, R03 as mandated by paragraph (g) of AD 2015-12-11, Amdt. 39-18186, Effective 07/23/15. |
| 2015-15-06 | 767-57A0076 | Inspect/Replace - Left and Right Trailing Edge Inboard Flap, Outboard Hinge Fitting Assembly . | 57-80124-03 | This EO accomplishes the intent of Boeing S/B 767-57A0076 R3, dated April 04, 2012 as mandated by AD 2015-15- 06, amendment 39-18215; effective date - November 13, 2015. Completion of Boeing S/B 767-57A0076 R3, dated April 04, 2012, PART 1 and PART 2 complies with AD 2015-15-06 para. (g), (h), & (n). Completion of Boeing S/B 767-57A0076 R3, dated April 04, 2012, PART 3, step 4, complies with AD 2015-15-06 para. (j) & (o). |
| 2015-21-09 | | Revision - Maintenance Inspection Program | 28-80155-01 | Within 30 days after the effective date of AD 2015-21-09, revise the maintenance or inspection program, as applicable, to add airworthiness limitation numbers 28-AWL-ENG, and 28-AWL-APU, by incorporating the information specified by this EO, E580155-01 and E580155-02 into the Airworthiness Limitations Section of the Instructions for Continued Airworthiness. |
| 2015-23-09 | Zodiac S/B MCF- SBU-35-001 | Inspect - Zodiac O2 Mask | 35-80220-01 | This EO complies with the intent of Zodiac Aerospace service bulletin MCF-SBU-35-001, Rev. 1 dated 12/03/2012 as mandated by paragraph (g) of AD 2015-23-09, amendment 39-18325, dated 12/28/2015. |
| 2015-26-07 | 767-53A0261 | Inspect/Replace - Vertical Stabilizer Attachment Fitting Barrel Nuts | 53-80125-02 | Compliance with this EO will accomplish an inspection and if necessary the replacement of the Vertical Stabilizer attachment fitting Barrel Nuts as call out in option 2, part 4 and 5 of Boeing S/B 767-53A0261, dated August 12, 2014 as mandated by AD 2015-26-07, dated February 4, 2016 paragraphs (g)(2) and (h). |
| 2016-04-07 | 767-53A0263 | Inspect - Skin & Splice Fittings at STR 29 STA 786 Ring Chord | 53-80160-02 | This EO addresses and complies with the intent of Boeing S/B 767-53A0263, R00, as mandated by AD 2016-04-07, Amdt. 39-18401, Effective 03/28/2016, Sections (g), (g)(1), and (g)(2). |
| 2016-25-01 | 767-22-0143 | Install - Automatic Stabilizer Trim On-Ground Inhibit System- Non Winglett Aircraft | 22-80191-01 | Accomplishing this task incorporates SB 767-22-0143 Revision 02 Dated 05/25/2016 as mandated by AD 2016-25-01 Par (c) (3) AMND 39-18727, Dated 01/31/2017. |
| 2016-25-01 | Aviation Partners SB AP767-22-005 767-22-0146 | Install - New Operational Program Software into the FCCs | 22-80178-01 | This EO satisfies the requirements of AD 2016-25-01, amendment 39-18727 effective January 31, 2017 by accomplishing Boeing SB 767-22-0146, Rev. 1, dated June 25, 2015 as mandated by Aviation Partners SB AP767-22-005, Rev. 1, dated June 16, 2015. |
| 2016-25-01 | 767-22-0143 | Test - Automatic Stabilizer Trim On-Ground Inhibit System | 22-80192-02 | Compliance with this EO incorporates the repeat inspections required by S/B 767-22-0143 R03, dated 06/09/2017 as mandated by paragraph (i) of AD 2016-25-01, amendment 39-18727 dated 01/31/2017. S/B 767-22-0143 R03, dated 06/09/2017 has been approved as an alternative method of compliance to the requirements of paragraph (i) of AD 2016-25-01. |
| 2016-25-29 | 767-25-0550 | Replace - Lower Cargo Compartment Insulation | 25-80236-01 | Accomplishing this EO complies with the intent of Boeing S/B 767-25-055, IR dated 01/30/2015 as required by paragraph (g) of AD 2016-25-29, amendment 39-18755 dated 01/31/2017 (for S/B Group 4 aircraft). Due to lack of prior operator's service bulletin accomplishment data, S/B Group/Configuration is assumed to be Group 4, Configuration 3. |
| 2017-19-05 | Siemens SIL, PMC- 26-002, Siemens SIL, PMC- 26-003 | Inspect/Replace - Siemens S.A.S. Smoke Detectors | 26-80234-01 | This EO complies with AD 2017-19-05, amendment 39-19035 dated 10/31/2017, paragraph (f). |
| 2018-07-18 | 767-53A0251 | Inspect - Fasteners, Section 41, (S-37L and S-37R) Lap Splice | 53-80257-01 | Compliance with this EO incorporates inspection and repair requirements of S/B 767-53A0251, rev. 1 dated 03/07/2017 as mandated by paragraph (g) and (h) of AD 2018-07-18, amendment 39-19249 dated 05/21/2018. AD 2018-07-18 supersedes AD 2015-19-12. |

| AD | S/B | Subject | EO | Statement of compliance |
|----------------------------------|--|--|-------------|---|
| 2018-11-14 | Aviation Partners Boeing Service Bulletin AP767-57- 010 Aviation Partners Boeing Service Bulletin AP767-57- 014 | Inspect - HFEC, Lower Wing Skin, External Surface - Stringers L-9.5 and L-6.5 | 57-80260-02 | Compliance with job card 57-80260-02-101 and -201 will accomplish the intent of APB S/B AP767-57-010, R11 dated 04/03/2017, part 1, as mandated by AD 2018-11-14, amendment 39-19302 dated 07/10/2018 paragraph (g). Compliance with job card 57-80260-02-301 and -401 will accomplish the intent of APB S/B AP767-57-014, R01 dated 04/12/2017, part 1, as mandated by AD 2018-11-14, amendment 39-19302 dated 07/10/2018 paragraph (h). |
| 2018-13-06 | Aviation Partners Boeing Service Bulletin AP767-57- 013 Aviation Partners Boeing DWG 767- 9420 | Inspect - HFEC, External Lower Wing Skin, Inboard Segment of Stringer L-9.5 | 57-80256-01 | Accomplishment of this EO complies with the intent of Aviation Partners Boeing Service Bulletin AP767-57-013, R01, Dated April 11, 2017, as mandated by paragraph (g) of AD 2018-13-06 amendment 39-19318 dated 08/22/2018. |
| 94-03-07, 94-16- 05, 99-27-05 | 767-32-A0116 767-32-0128 767-32-0152 | Inspect - Brake Assemblies | 32-80252-01 | Accomplishing this EO verifies compliance with the following ADs by verifying that S160T300-() carbon brakes are NOT installed on the aircraft. Boeing S/B 767-32-A0116 R02 dated December 14, 2000 as mandated by AD 94-03-07, amendment 39-8814 dated February 18, 1994. Boeing S/B 767-32-0128 R03 dated December 14, 2000 as mandated by AD 94-16-05, amendment 39-8995 dated September 14, 1994. Boeing S/B 767-32-0152 R02 dated July 10, 1997 as mandated by AD 99-27-05, amendment 39-11486 dated February 8, 2000. |
| N/A | None | Inventory - Avionics Equipment, Post Cargo Mod | 25-80189-02 | None |
| | 767-27A0167 767-27-0196 | Inspect - Right Wing Trailing Edge Flap System 3-10 Links | 27-80259-01 | This EO complies with Part 3 of Boeing S/B 767-27A0167, Revision 3, dated December 19, 2007, for Group 1 airplanes, and part 1 of Boeing SB 767-27-0196 Revision 00, dated April 21, 2005 as mandated by AD 2006-10-08, amendment 39-14592, effective June 19, 2006, paragraph (i)(1) or (j). S/B 767-27A0167, Revision 3 has been approved as an AMOC to the requirements of AD 2006-10-08. S/B 767-27-0196 is terminating action for the repetitive inspections of AD 2006-10-08 (g), (h) and (i). This EO is for the Right Wing only, EO 27-80258-01 inspects the Left Wing links. |
| | 767-76-0010 | Inspect - Thrust Control Cables, CF6-80A | 76-80245-01 | This EO complies with the repetitive inspections mandated by paragraph (a) of AD 1999-21-18, amendment 39- 11363 dated 11/22/1999. |
| | 767-27A0064 | Inspect - Flight Control Fastener Run-off Torque | 27-80251-01 | Completion of this EO accomplishes the intent of Boeing Service Bulletin 767-27A0064 Rev 02, dated April 25, 1986 as mandated by paragraphs (a) and (b) of AD 85-21-51, amendment 39-5232 dated February 24,1986. |
| | 767-27A0167 767-27-0196 | Inspect - Left Wing Trailing Edge Flap System 3-10 Links | 27-80258-01 | This EO complies with Part 3 of Boeing S/B 767-27A0167, Revision 3, dated December 19, 2007, for Group 1 airplanes, and part 1 of Boeing SB 767-27-0196 Revision 00, dated April 21, 2005 as mandated by AD 2006-10-08, amendment 39-14592, effective June 19, 2006, paragraph (i)(1) or (j). S/B 767-27A0167, Revision 3 has been approved as an AMOC to the requirements of AD 2006-10-08. S/B 767-27-0196 is terminating action for the repetitive inspection requirements of AD 2006-10-08 (g), (h) and (i). This EO is for the Left Wing only, EO 27-80259-01 inspects the Right Wing links. Not |